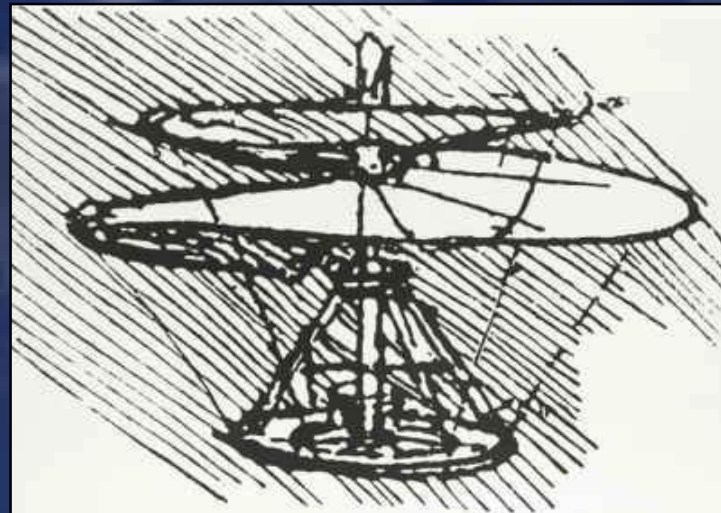


Main Rotor Blade Analysis in Helicopter Accident Investigation

by Sam Webb

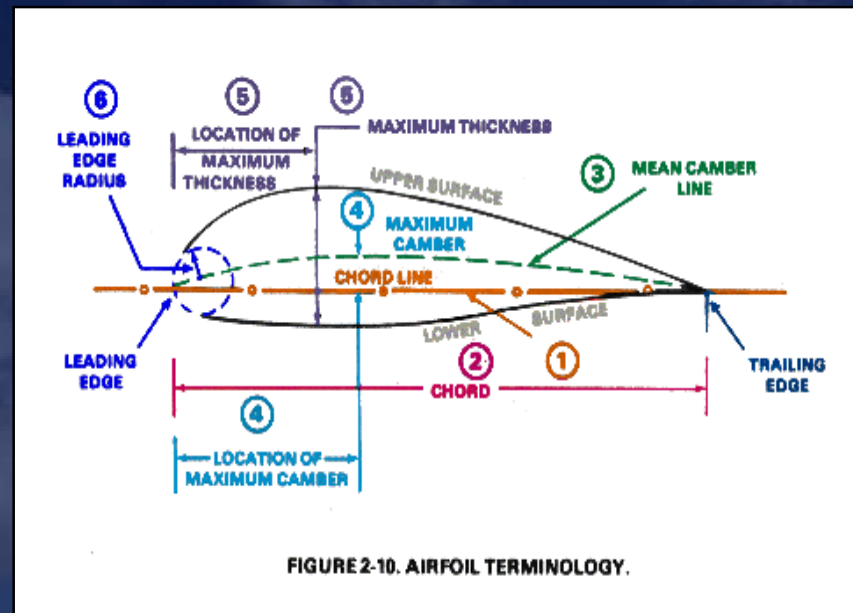


Purposes and Goals

- *Establish the last flight path, heading, and attitude and of the helicopter*
- *Establish the RPM of the main rotor and engine power levels*
- *Establish drive train continuity*
- *Establish sequence of events of the accident*

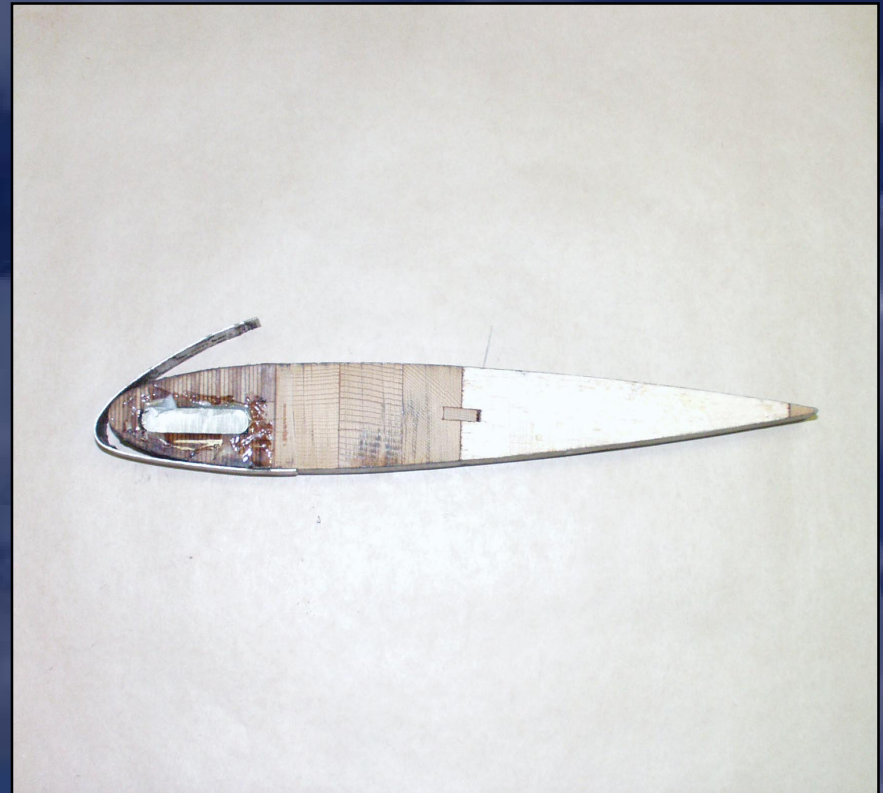
Design of Main Rotor Blades

→ Investigator must have a limited knowledge of the design of the rotor system and blade



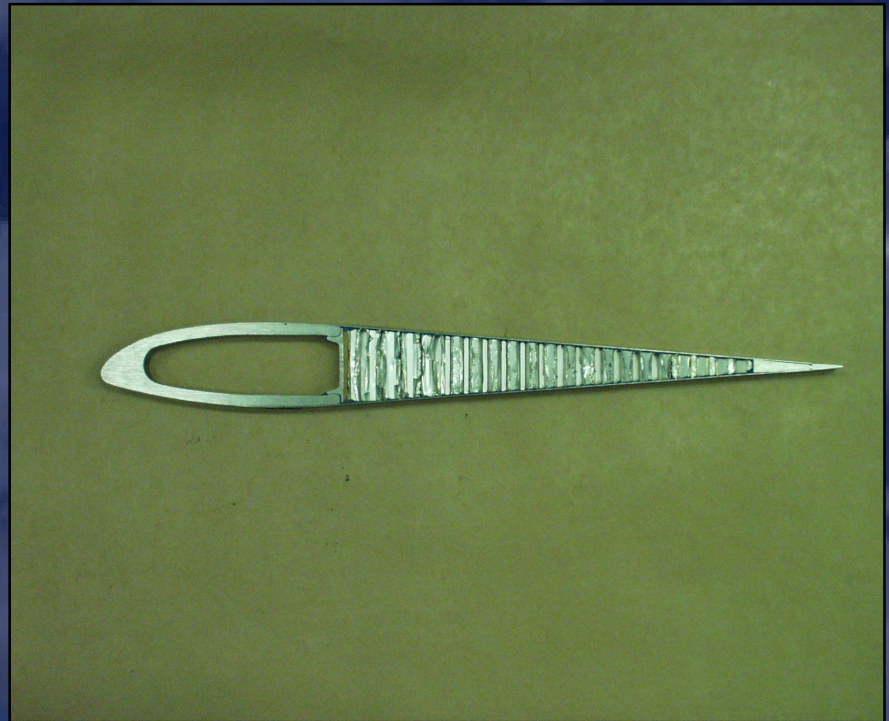
Design of Main Rotor Blades

- Wooden symmetrical airfoil
- Nickel leading edge, wooden filler, steel spar



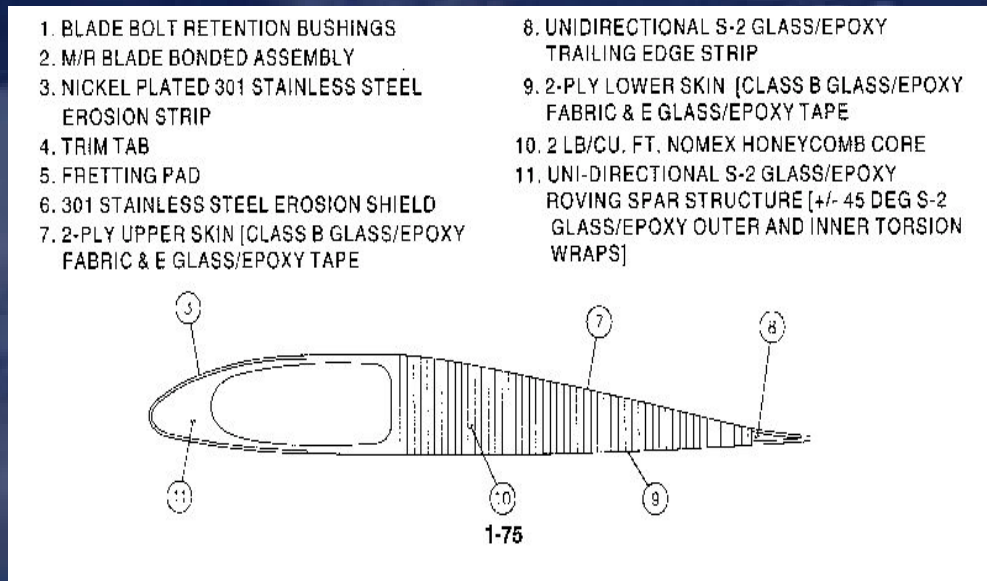
Design of Main Rotor Blades

→ Aluminium spar and skin, honeycomb core



Design of Main Rotor Blades

- Composite asymmetrical airfoil
- Uni-directional S-2 glass/ epoxy fabric & E glass/ epoxy tape, fibreglass spar
- Superior fatigue tolerance, notch and corrosion resistance



Main Rotor Design Differences

- Normally wider cord / longer length blades
- Higher main rotor RPM/ higher blade inertia

**Two Bladed Systems,
(teetering, semi-rigid)**



Main Rotor Design Differences

- Blades usually lighter construction, less cord, less length
- Main rotor RPM values marginally lower to compensate for cord area differences

Fully articulated, multi-blade systems



Main Rotor Design Differences

- Blades same construction as multi-bladed system
- Main rotor RPM values same as multi-bladed system

Coaxial- counter rotating, multi-blade systems



Main Rotor Damage Differences

- Blade bending, not shattering
- Blades generally intact
- Blade bending inboard and downward
- High angle of attack = blade tearing aft of spar



Low Main Rotor RPM

Main Rotor Damage Differences

→ Blade tip weights intact



Weights

Low Main Rotor RPM

Main Rotor Damage Differences

- Spar fractures, trailing edge separation
- Honeycomb fractured and separated
- Damage mainly to outboard sections of blade



Fractured spar

High Main Rotor RPM

Main Rotor Damage Differences

- Tip weights “ejected”
- May travel for kilometres from the impact site (farther distances for multi-bladed systems)



High Main Rotor RPM

Main Rotor Damage Differences

- First blade to impact surface most damaged
- Following blades exhibit lesser damage due to main rotor inertia bleed off



High Main Rotor RPM

Main Rotor Damage Differences

→ Damage is in plane



Shattering

High Main Rotor RPM

Main Rotor Damage Differences

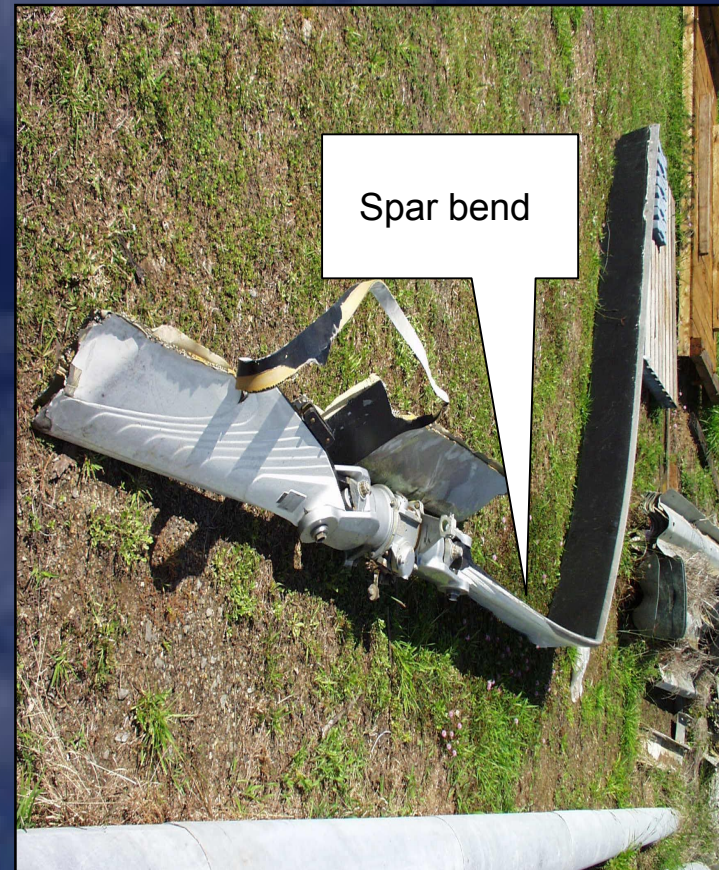
- Massive blade distortion on multi-bladed systems
- Blades sustain more damage due to lighter construction
- Articulated systems, dynamic stops fail



High Main Rotor RPM

Main Rotor Damage Differences

- Blade spindling/
distortion
- Spar fractures
depending on main
rotor RPM
- Bending of spar
depending on angle of
attack



Damage From Water Impact

Factors Effecting Main Rotor RPM

- Inertia of the blade- with high inertia will lose RPM slowly with increased angles of attack
- Higher the helicopter gross weight and/or density altitude= more the blade wants to overspeed
- Manoeuvring- tends to increase RPM due to energy enhancement of the rotor system

Supporting Evidence

- Will collaborate other evidence
- Distorted or broken
- Direction of force will collaborate power on or off and autorotation



Main Rotor Controls

Supporting Evidence

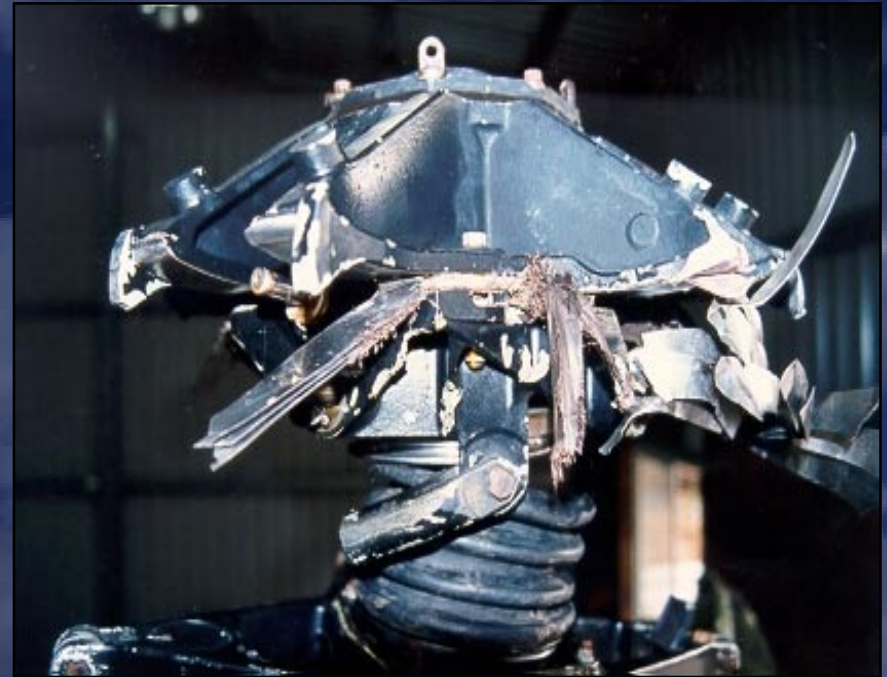
→ Damage to pitch horn, blade grip, mast assembly on semi-rigid and rigid systems = indication of high RPM



Main Rotor System

Supporting Evidence

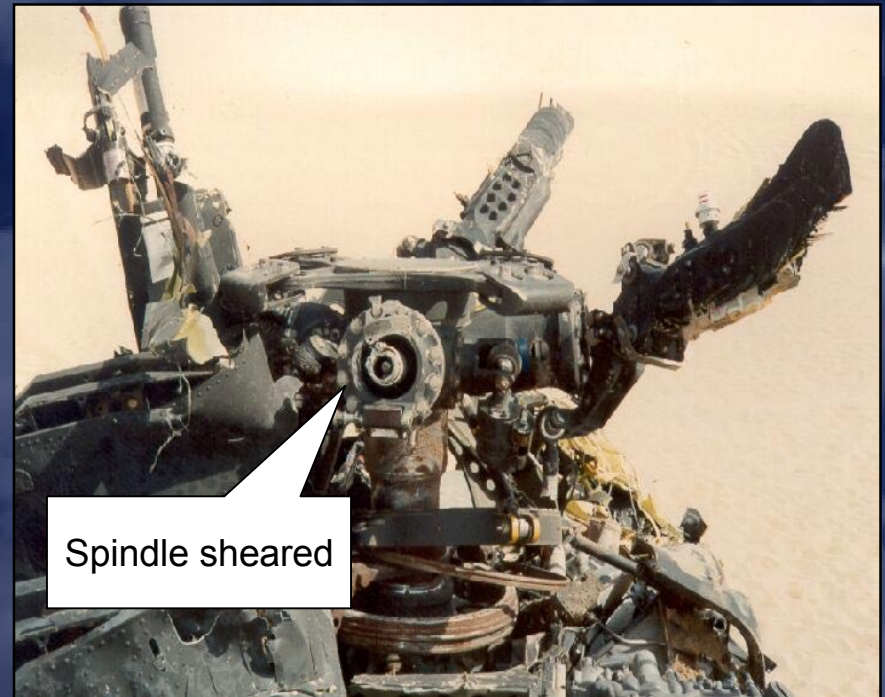
→ Torque-tension strap
damage



Main Rotor System

Supporting Evidence

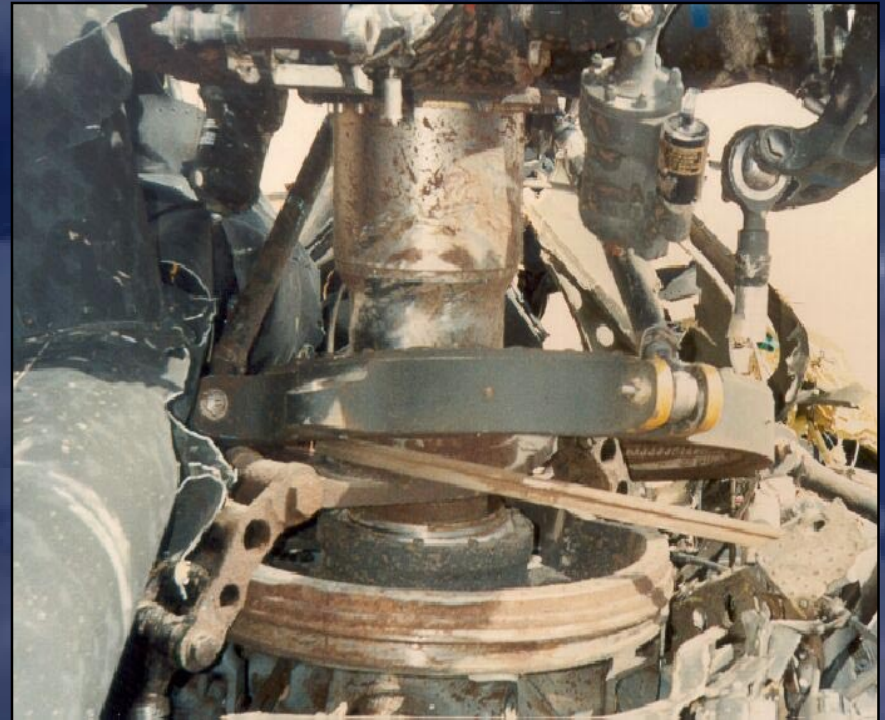
→ MR Spindle shear damage



Main Rotor System

Supporting Evidence

→ Swashplate duplex bearing damage = indication of high RPM



Main Rotor Swashplate

Supporting Evidence

- Pitch change rod end attachment damage
- Static overload indications= high RPM
- “Necking” indications= low RPM



Main Rotor Controls

Summary

Sometimes you just
have a bad day!



Summary

Sometimes you never understand!?!?!?!?!?





Thank you for your attention!