



The Australian Transport Safety Bureau

The ATSB aims to maintain and improve transport safety through excellence in:

- 'no blame ' independent transport accident, incident and safety deficiency investigation,
- safety data, research, and pro-active systemic analysis,
- communication and education, and
- leading development of national and international safety strategies.

AIRCRAFT AVIONICS WIRING & ELECTRICAL SYSTEMS

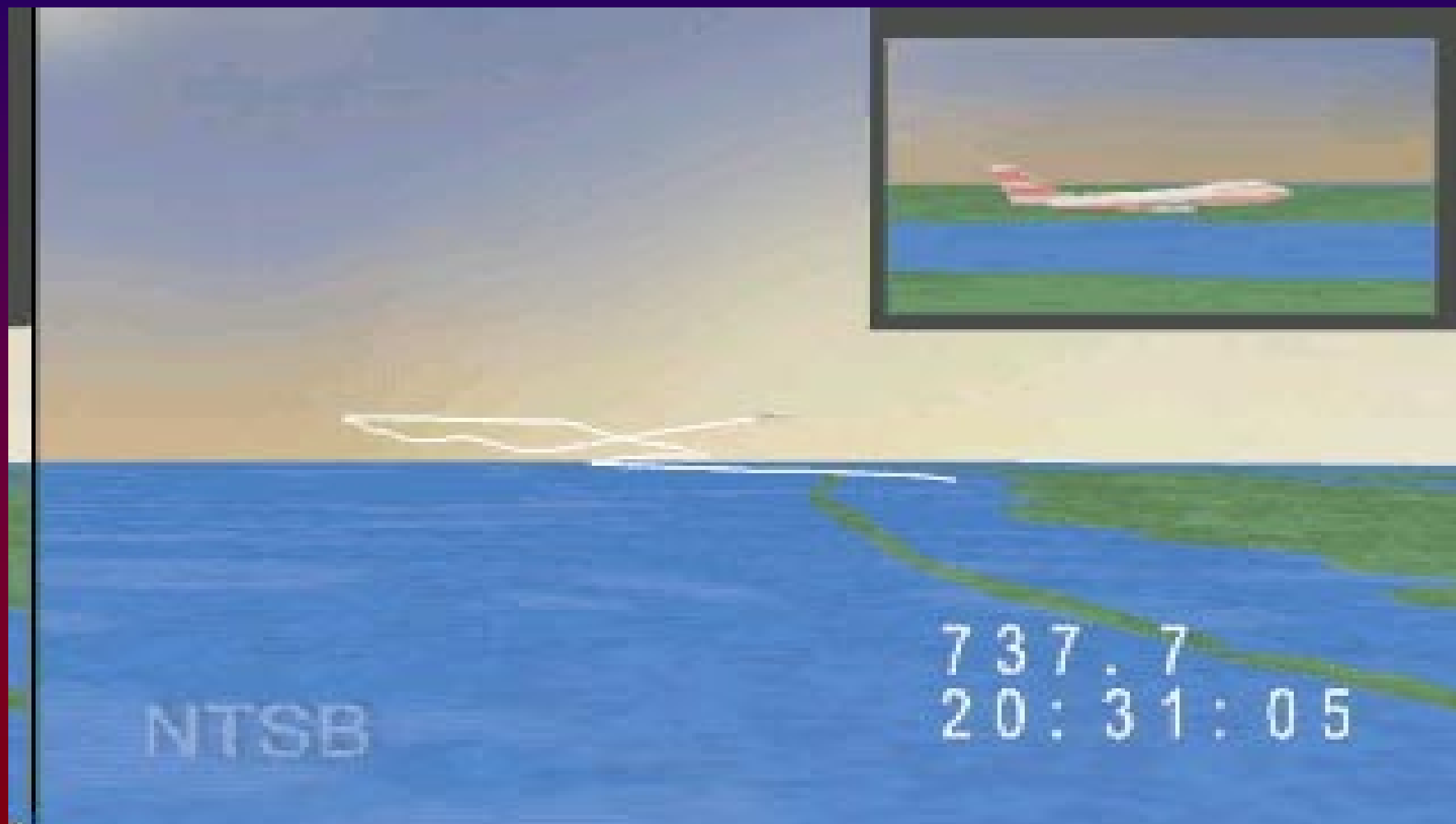
TWA 800
230
LIVES LOST

-
- 1. HOT DAY WITH DELAYS
ON TAKE OFF***
 - 2. LOW FUEL QUANTITY IN
CENTRE WING TANK***
-

***3. HEATING OF THE LOW
QUANTITY OF FUEL BY THE AIR
CONDITIONER PACKS
LOCATED UNDER THE CENTRE
WING TANK, CREATING A
VOLATILE FUEL AIR MIXTURE.***

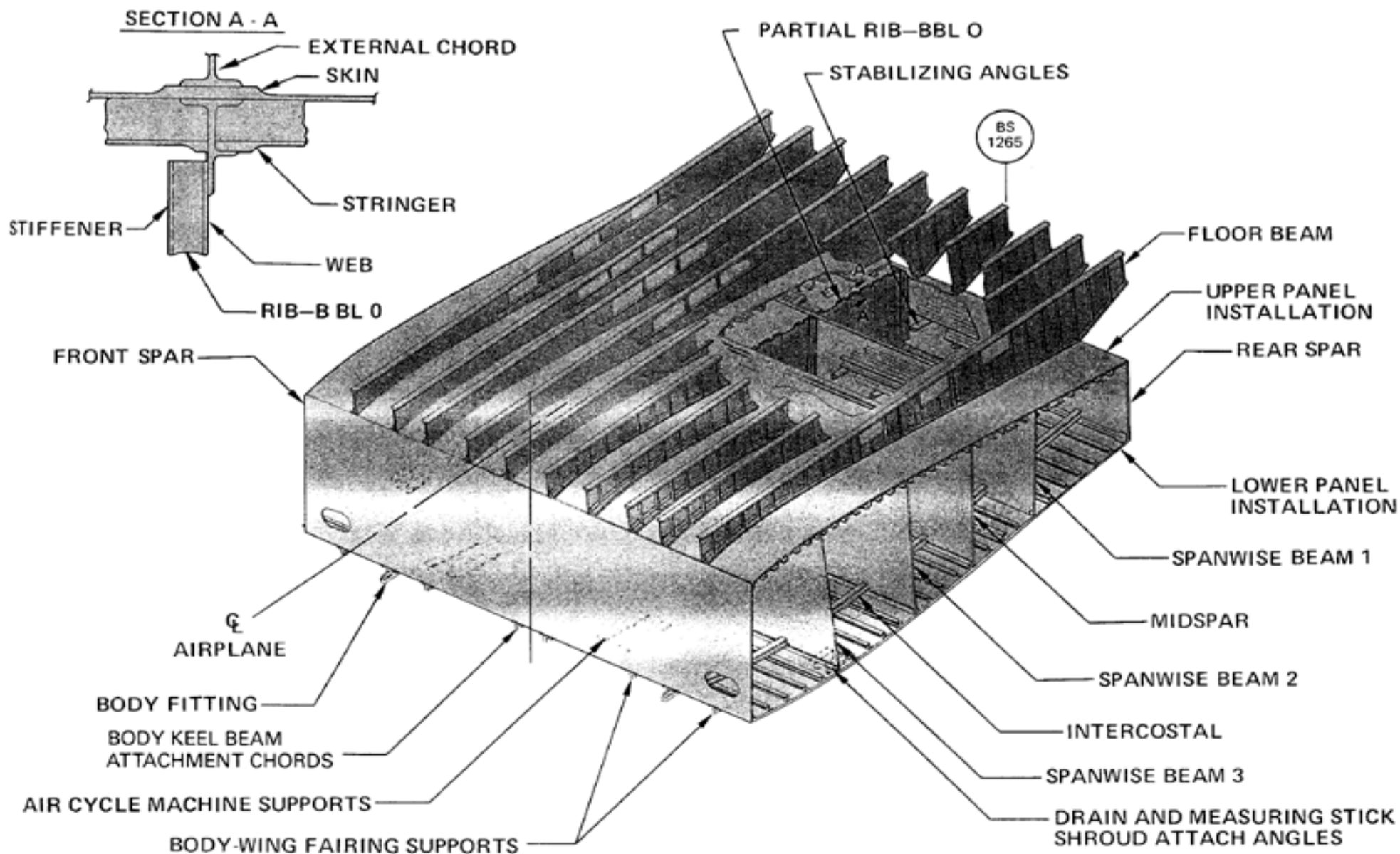
4. IGNITION SOURCE

5. EXPLOSION



CENTRE WING TANK

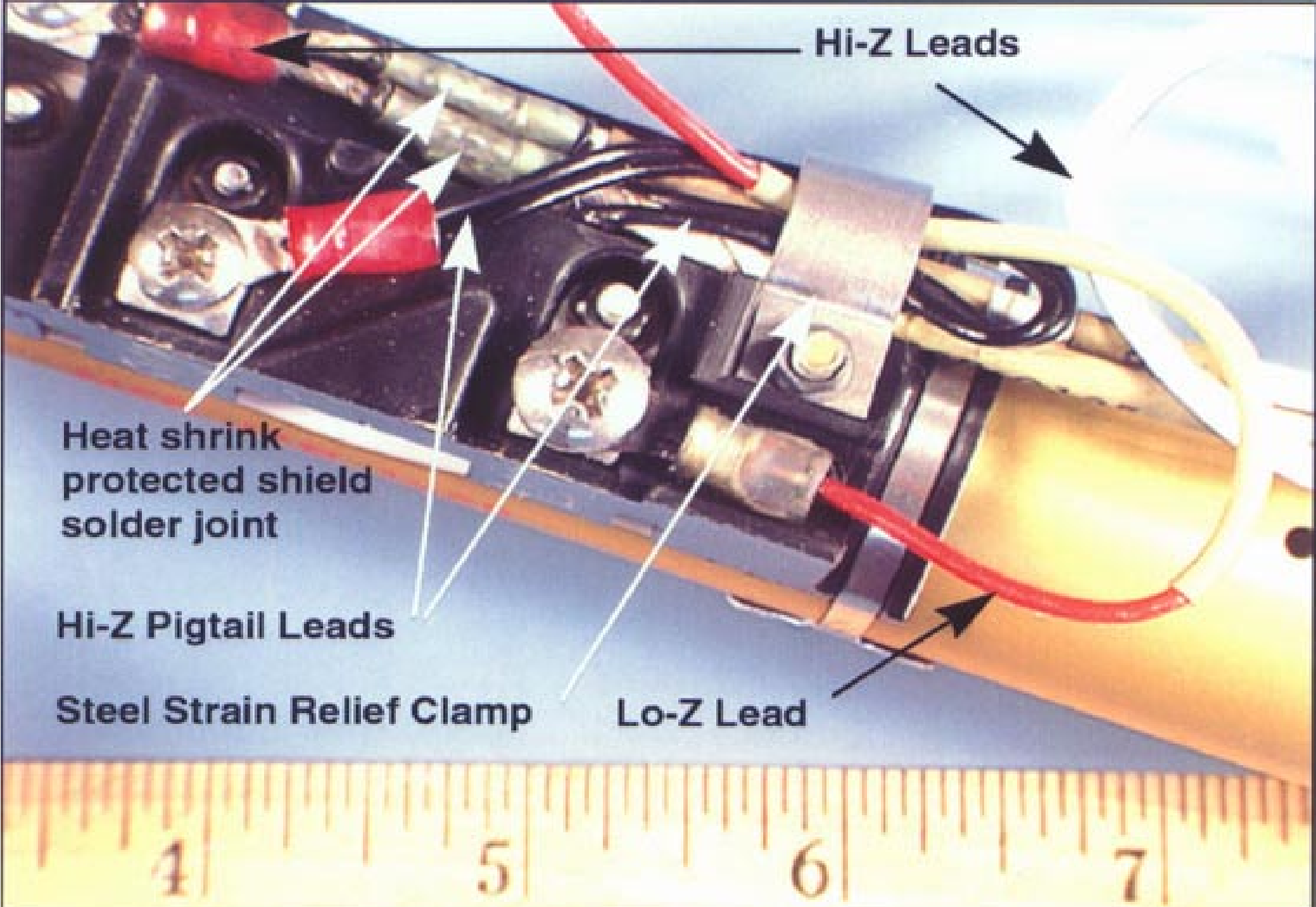
Wing Center Section







FUEL QUANTITY PROBE



Hi-Z Leads

Heat shrink
protected shield
solder joint

Hi-Z Pigtail Leads

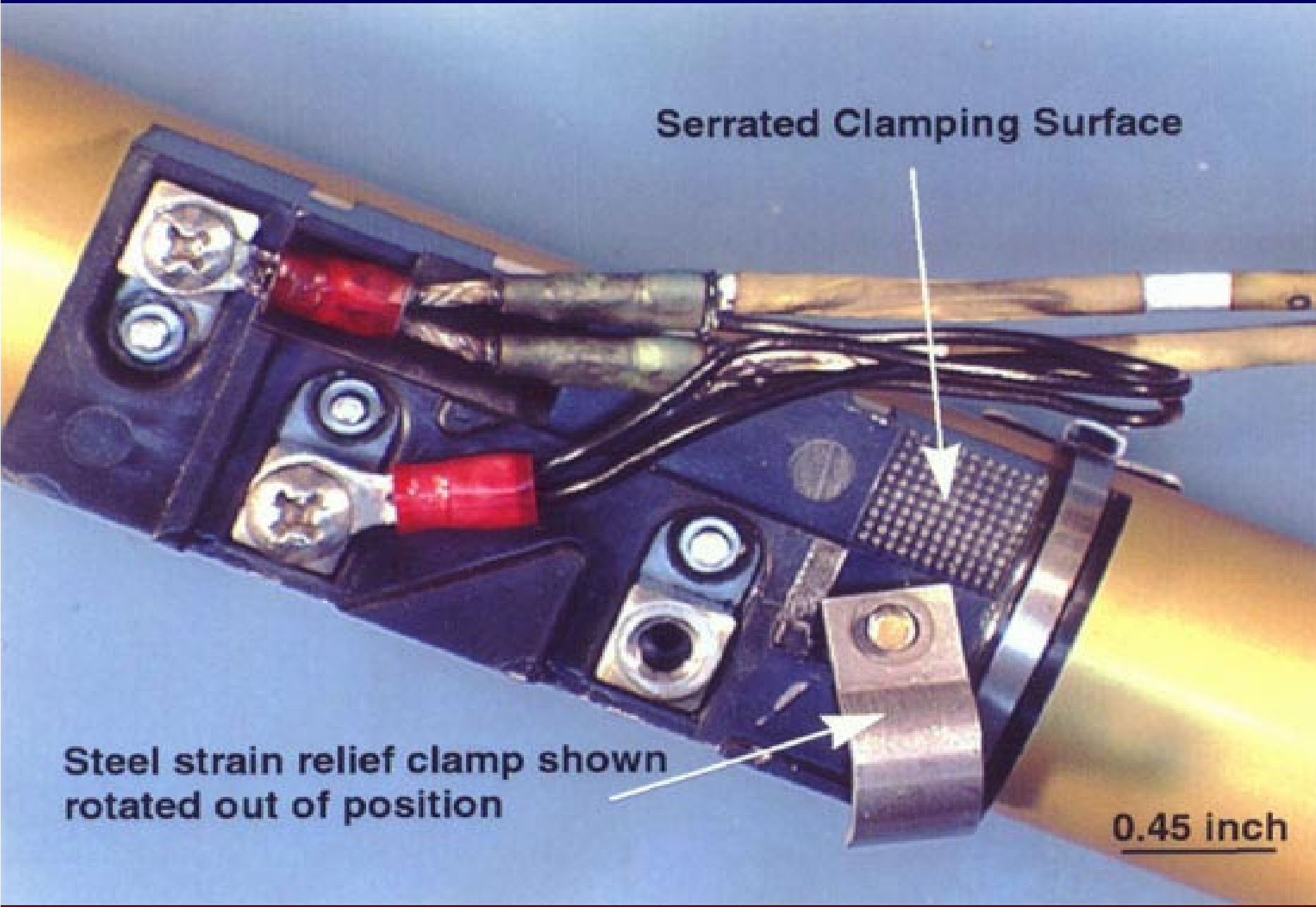
Steel Strain Relief Clamp

Lo-Z Lead

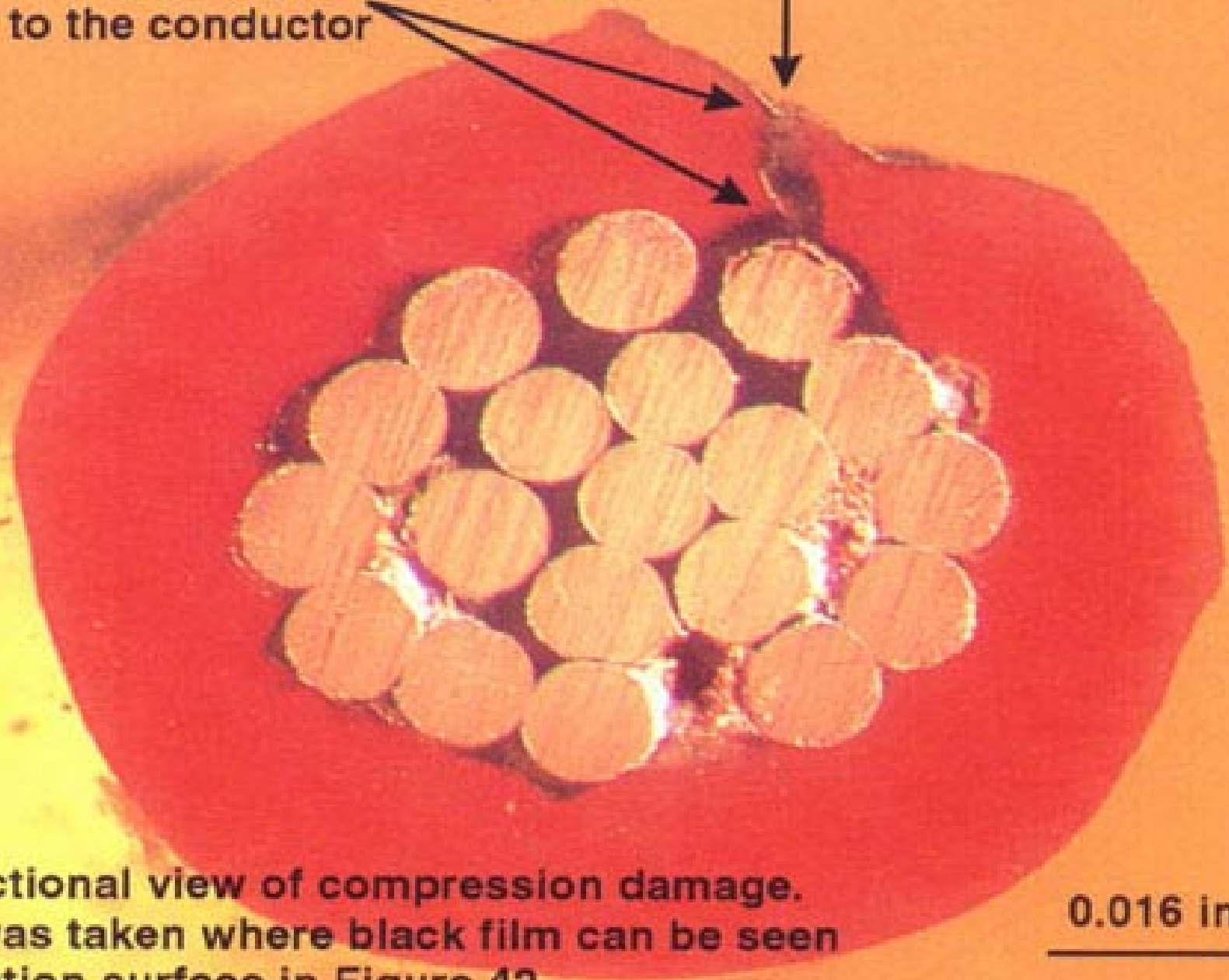
Serrated Clamping Surface

**Steel strain relief clamp shown
rotated out of position**

0.45 inch



PTFE insulation thinned and split
through to the conductor



Cross-sectional view of compression damage.
Section was taken where black film can be seen
the insulation surface in Figure 42

0.016 inch

***THE NATIONAL
TRANSPORTATION SAFETY
BOARD DETERMINED THAT THE
PROBABLE CAUSE OF THE
TWA800 ACCIDENT WAS AN
EXPLOSION OF THE CENTRE
WING TANK***

***THE SOURCE OF THE IGNITION
ENERGY FOR THE EXPLOSION
COULD NOT BE DETERMINED
WITH CERTAINTY***

***BUT, OF THE SOURCES
EVALUATED BY THE
INVESTIGATION, THE MOST
LIKELY WAS A SHORT
CIRCUIT OUTSIDE THE
CENTRE WING TANK.***

***ALLOWING EXCESSIVE
VOLTAGE TO ENTER THE
CENTRE WING TANK
THROUGH ELECTRICAL
WIRING ASSOCIATED WITH
THE FUEL QUANTITY
INDICATION SYSTEM (fuel probe)***

CONTRIBUTING FACTORS DESIGN & CERTIFICATION

- THE CONCEPT THAT FUEL TANK EXPLOSIONS COULD BE PREVENTED SOLELY BY PRECLUDING ALL IGNITION SOURCES***
 - THE BOEING 747 HAS HEAT SOURCES LOCATED BENEATH THE CENTRE WING TANK WITH NO MEANS TO REDUCE THE HEAT TRANSFERRED TO THE CENTRE WING TANK OR TO RENDER THE VAPOR IN THE TANK NONFLAMMABLE***
-

NTSB RECOMMENDATIONS

1. NITROGEN INERTING SYSTEM

***2. MONITORING OF FUEL
TEMPERATURES***

***3. FUEL TANK TEMPERATURE
PROBES AND COCKPIT DISPLAYS***

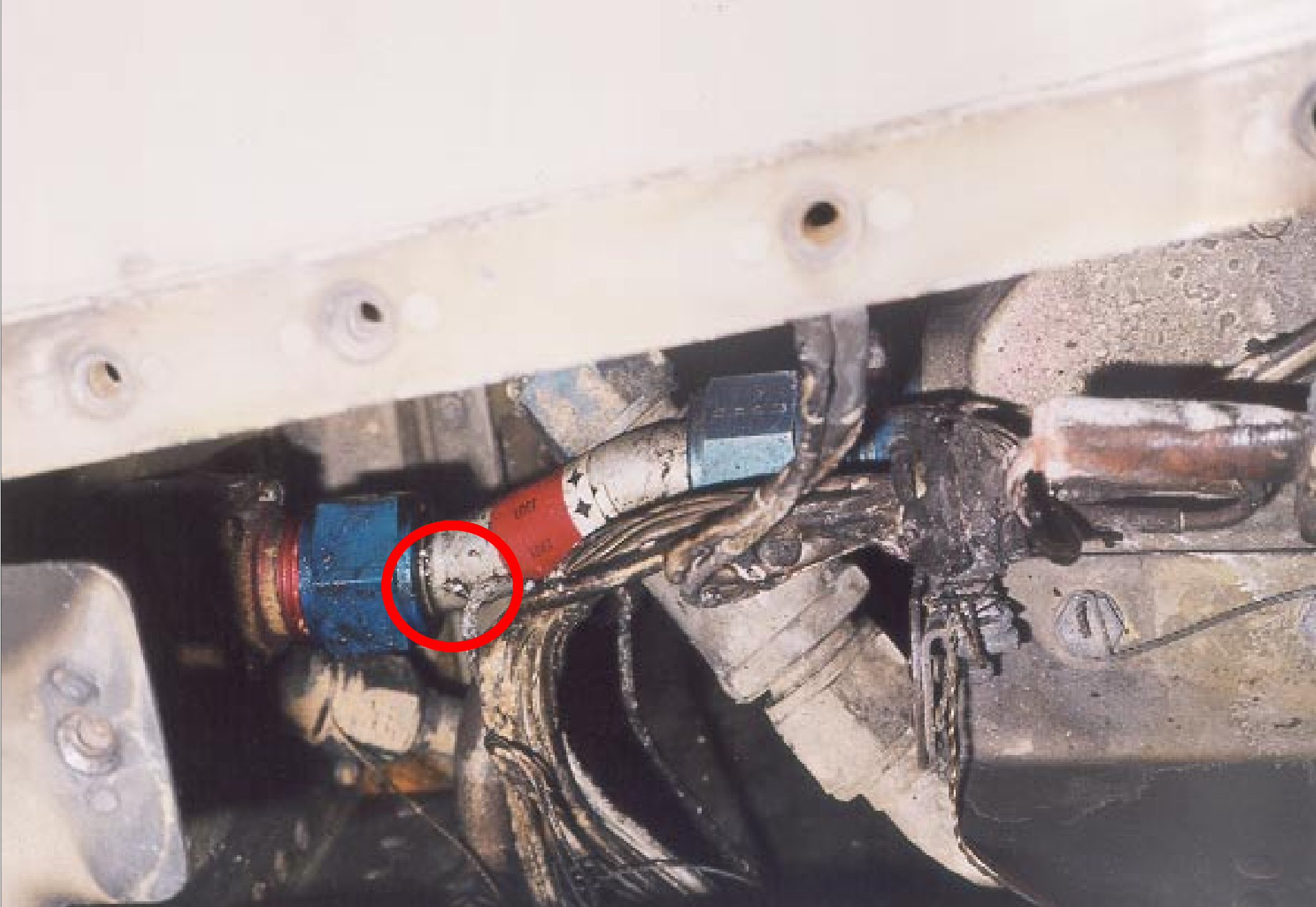
BEECHCRAFT 1900D WING FIRE

***THE INVESTIGATION FOUND
EVIDENCE OF A FUEL FED FIRE IN
THE AFT AREA OF THE RIGHT
MAIN LANDING GEAR WHEEL
WELL AND IN THE WING
IMMEDIATELY OUT BOARD OF THE
RIGHT ENGINE NACELLE***

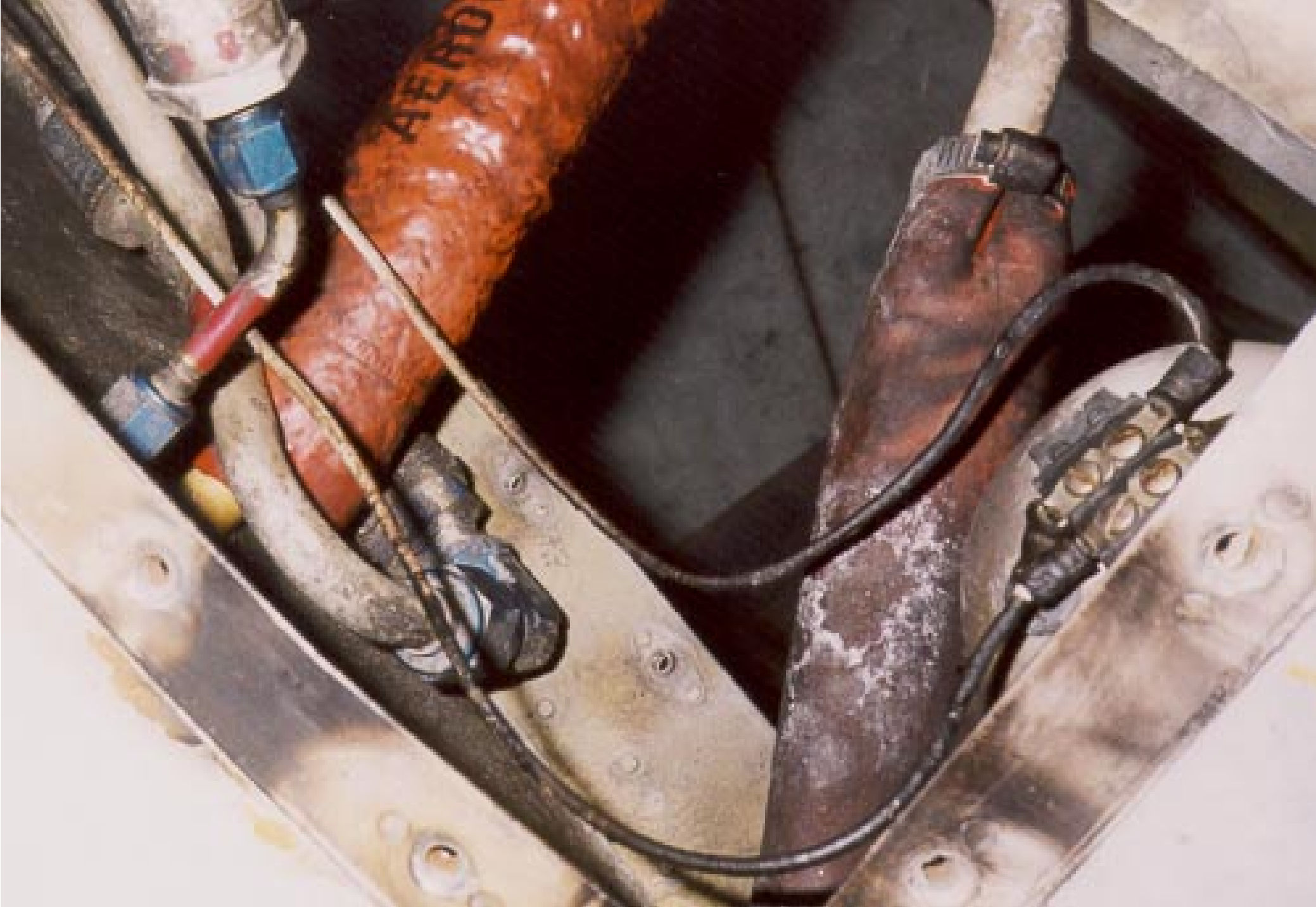




***FUEL TO THE FIRE HAD BEEN
SUPPLIED FROM TWO
DAMAGED ALUMINIUM ALLOY
FUEL TANK LINES IN THE
RIGHT WING EQUIPMENT BAY***



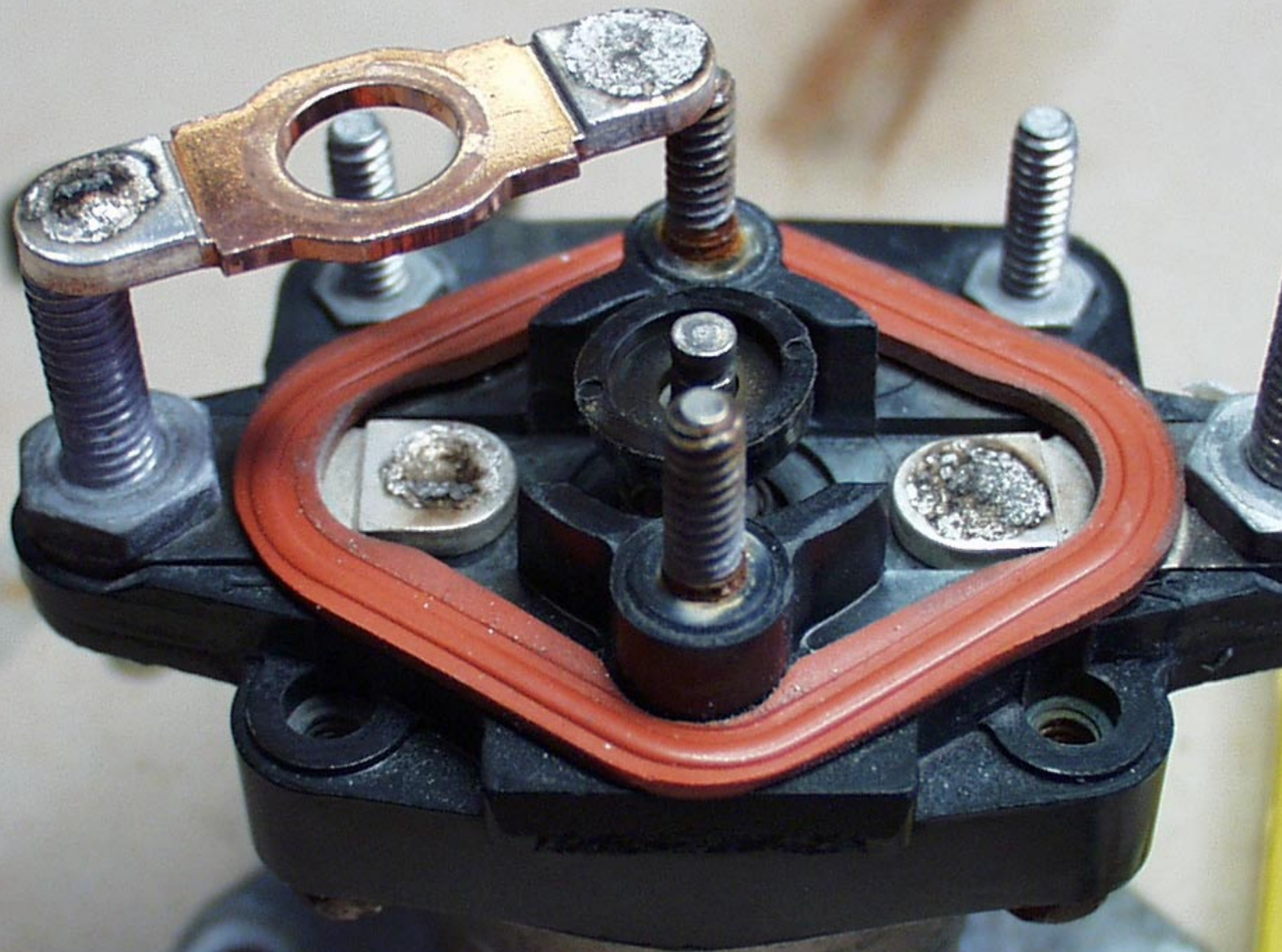




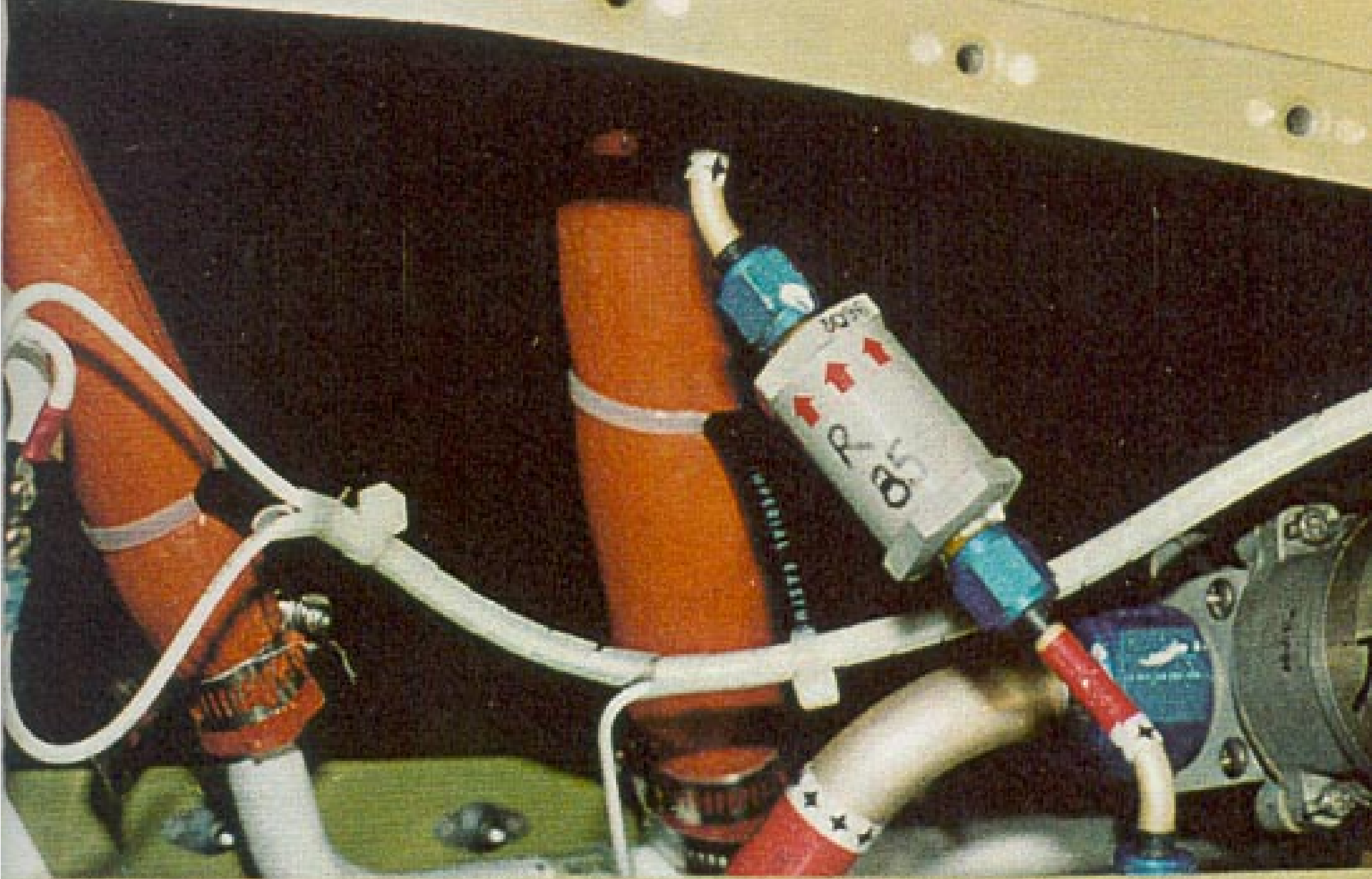








***THE AIRCRAFT INVOLVED IN
THIS ACCIDENT HAD BEEN
REPAIRED PRIOR TO THE
ACCIDENT AND THE
CORRECT WIRING STAND -
OFFS WERE NOT INSTALLED***





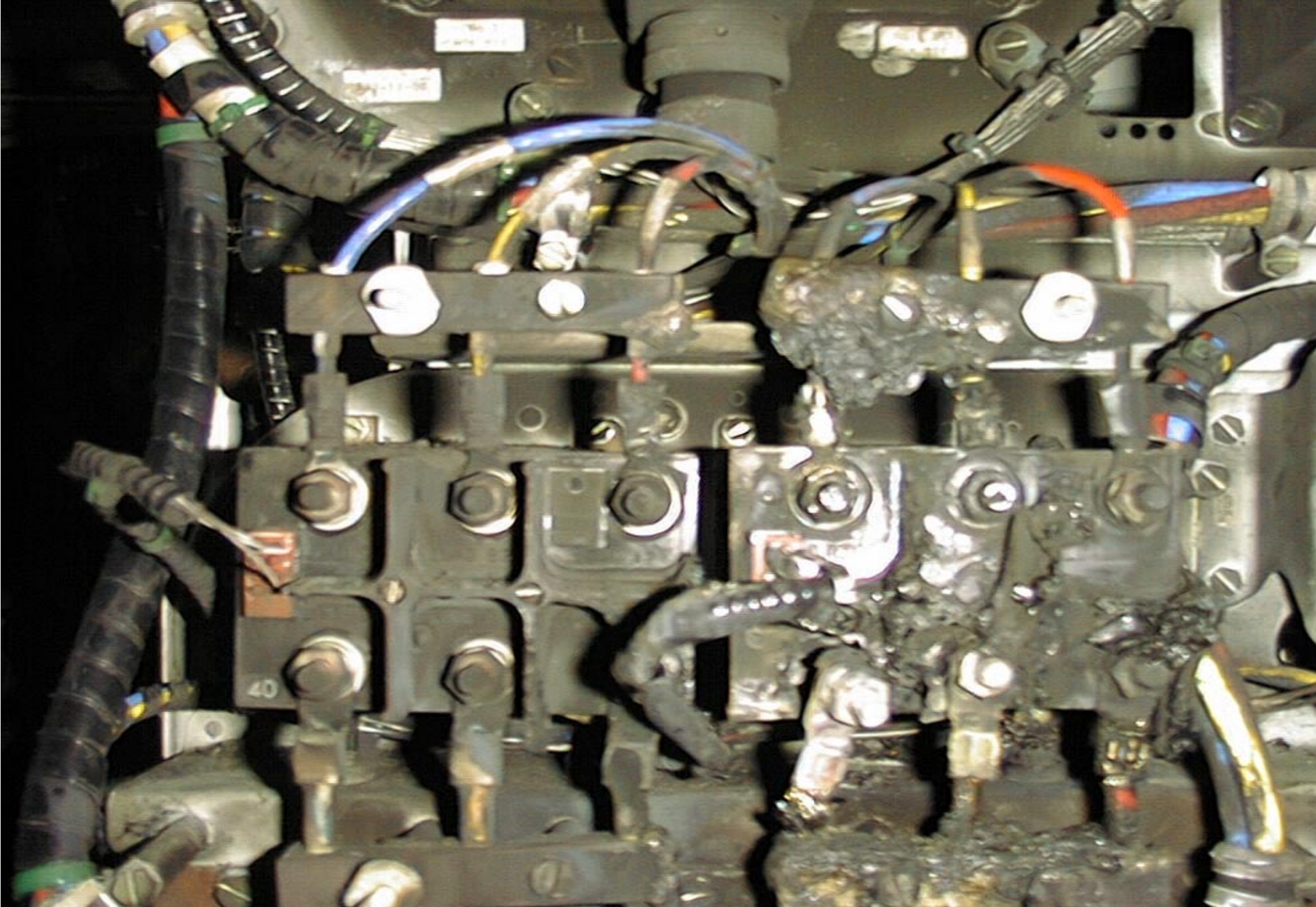
INTERIM RECOMENDATION
ALL OPERATORS INITIATE
IMMEDIATE WIRING AND
FUEL LINE INSPECTION IN
WING ZONES 531 & 631
CLOSED AND ACCEPTED

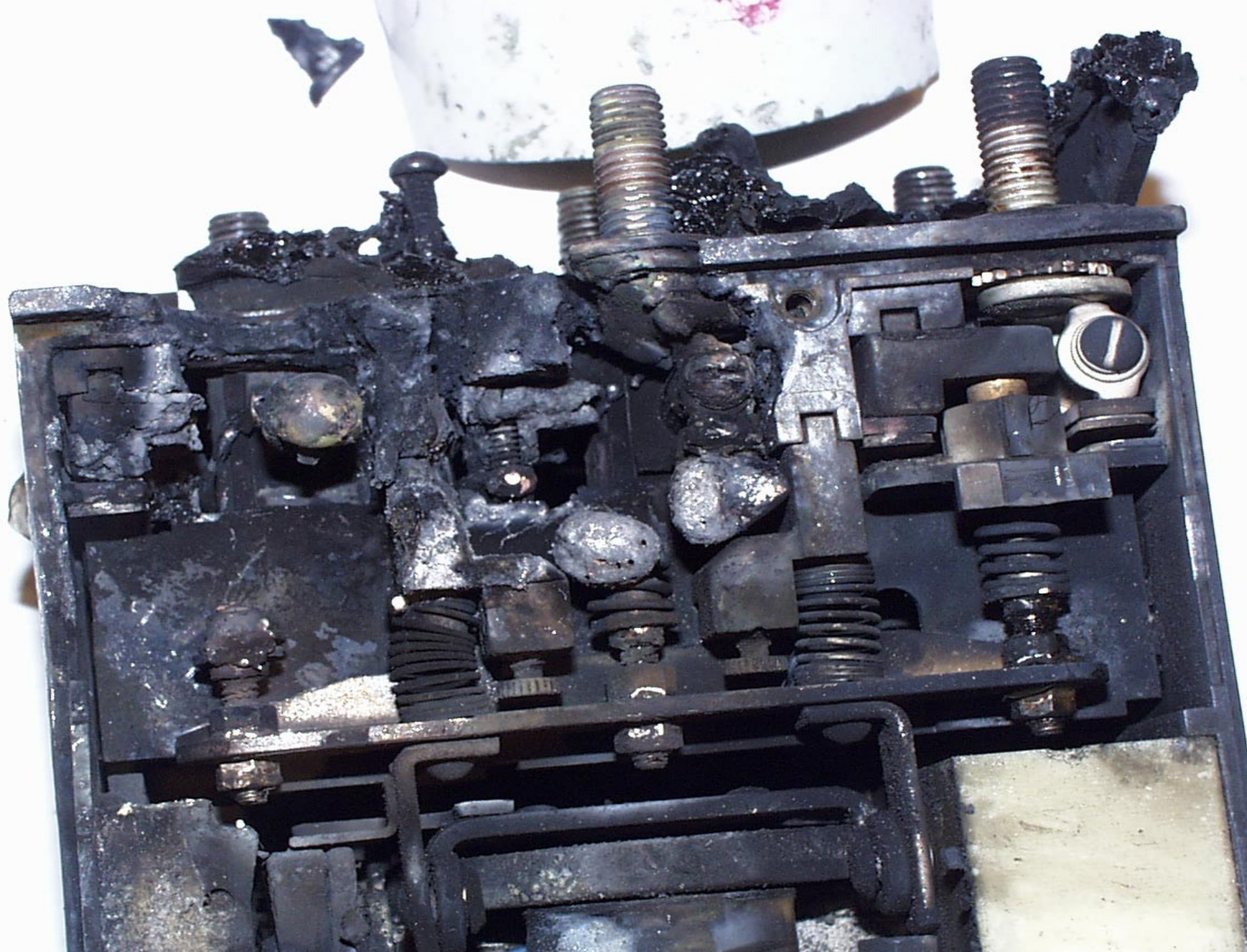
BAe146 REMOTE CONTROL CIRCUIT BREAKER (RCCB) FIRE

***HIGH CURRENT DRAW
DEVICES SUCH AS A
HYDRAULIC PUMP REQUIRE
SWITCHING DEVICES (RCCB)
OR RELAYS***

***THESE DEVICES REQUIRE
REGULAR MAINTENANCE
INSPECTIONS PARTICULARLY IN
AGEING AIRCRAFT AS THESE
DEVICES ARE DEGRADED BY
THE HEAT PRODUCED BY THEIR
OWN OPERATION***







***THE RCCB LOCATED IN THE
EQUIPMENT BAY WITHIN THE
CABIN IS NOT MONITORED BY FIRE
OR SMOKE DETECTORS THE
TECHNICAL CREW WERE ALERTED
BY SMOKE IN THE COCKPIT,
SYSTEM FAILURES AND THE FIRE***

***THE AIRFRAME MANUFACTURER'S
FAILURE TREND DATA FOR THE
RCCB WAS EXAMINED AND IT WAS
FOUND TO BE A VERY RELIABLE
COMPONENT CONSEQUENTLY A
RECURRENCE OF THIS TYPE
UNLIKELY***

SAFETY ACTION

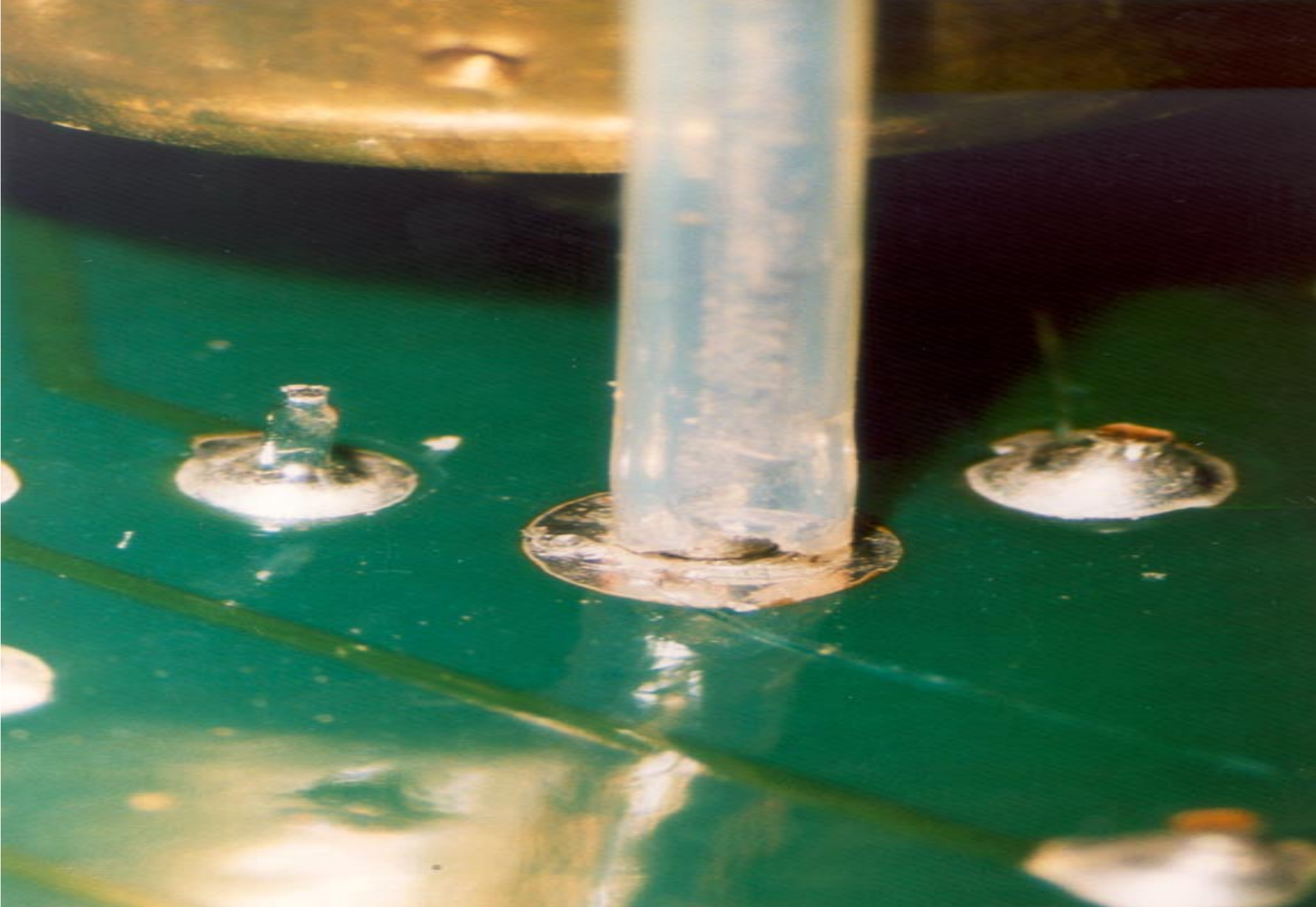
***AFTER CONSULTATION WITH THE
ATSB THE OPERATOR INSTIGATED
A NEW PROCEDURE TSI-146-24-004***

-
- ***FITMENT OF HEAT SENSITIVE DECALS TO THE RCCB***
 - ***RECORDING THE OPERATING TEMPERATURES IN A DATA BASE TO ANALYSE TEMPERATURE TRENDS THIS METHOD TURNED UP ANOTHER FAULTY RCCB (PROACTIVE)***
-

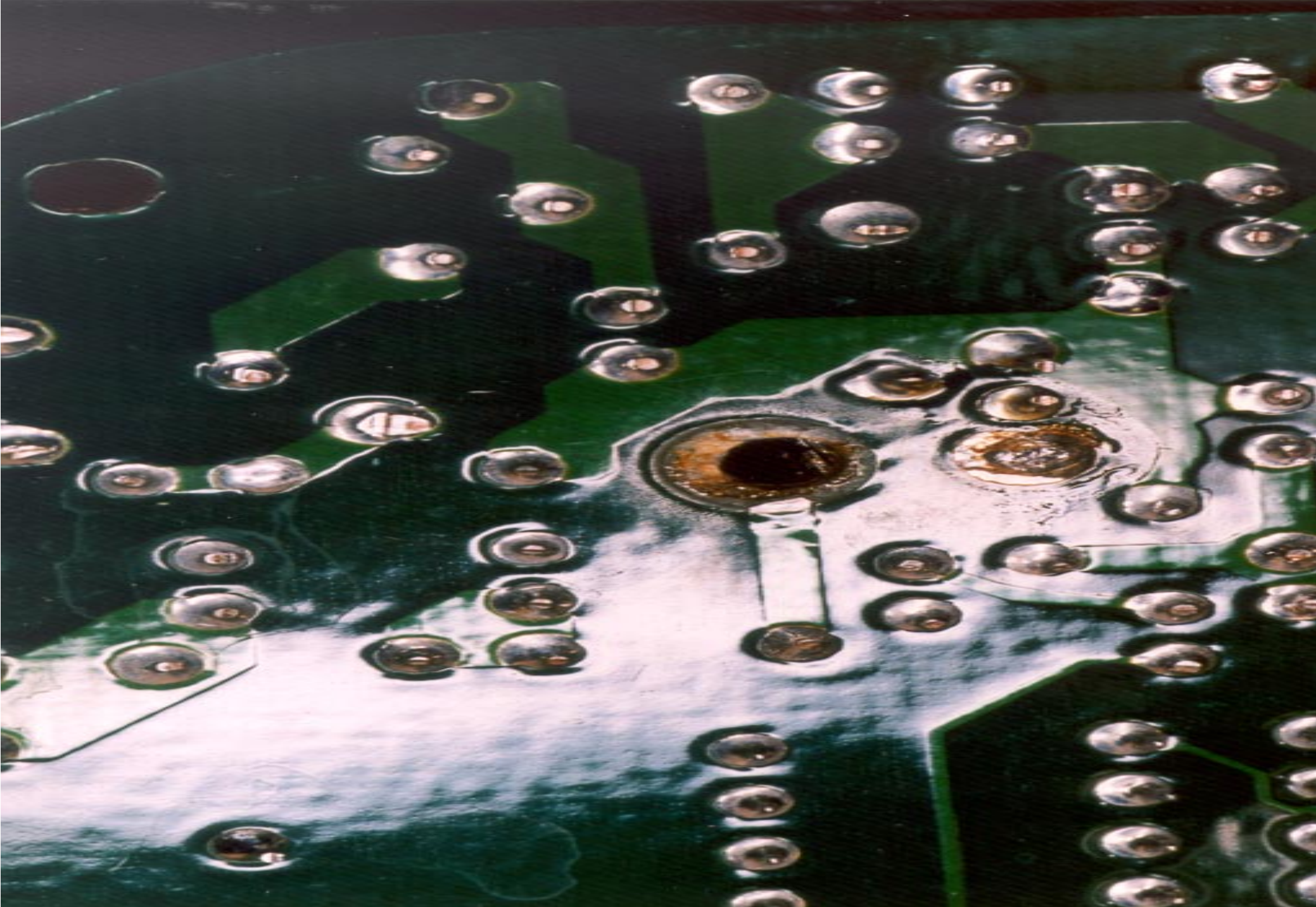
EMERGENCY LOCATOR BEACONS (ELT'S)

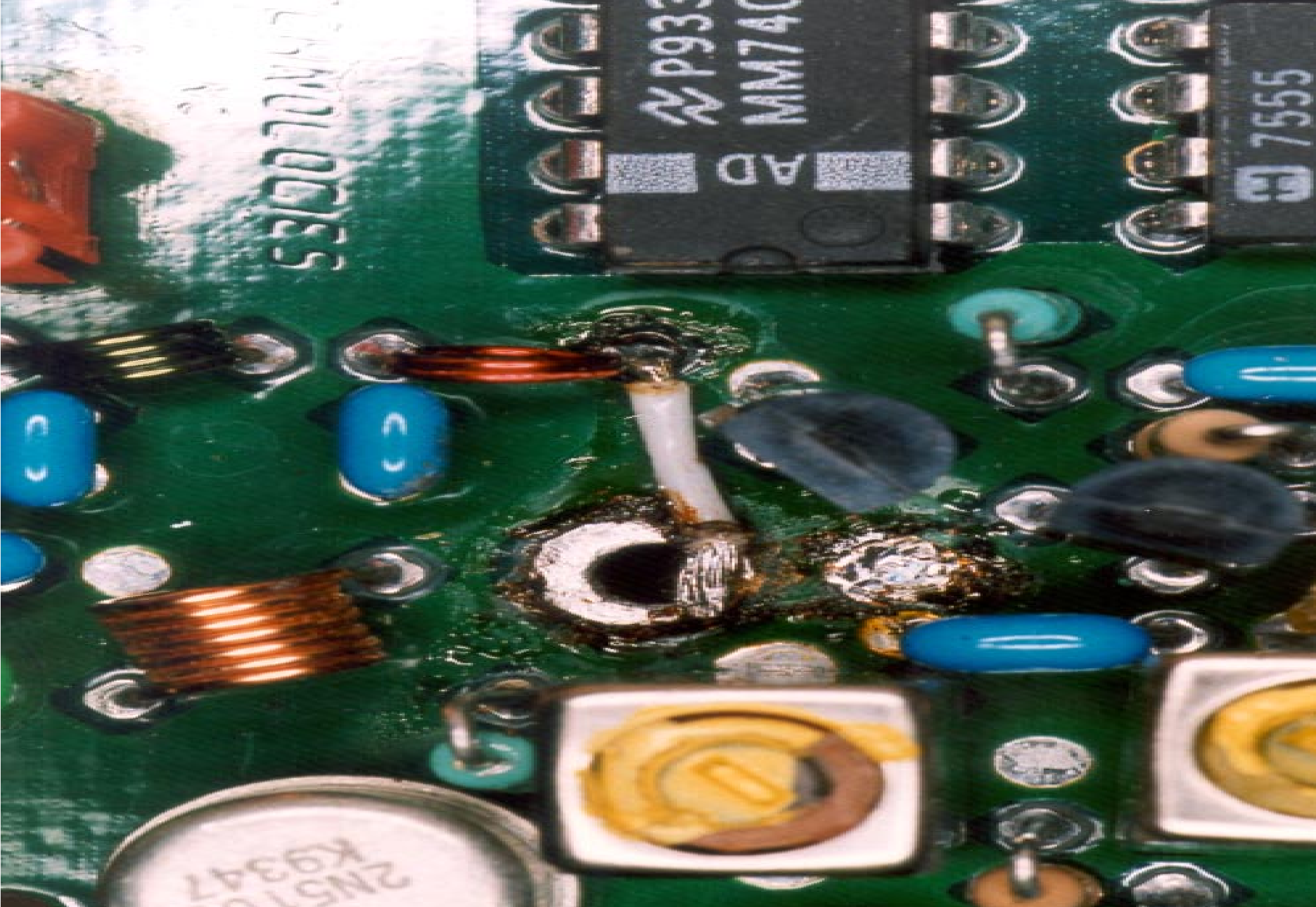
***EXAMINATIONS OF ELT'S
RETURNED FROM ACCIDENT
SITES HAVE REVEALED
FAILURES OF DIFFERENT TYPES
IN THIS CASE THE UNIT DID NOT
TRANSMIT A SIGNAL DUE TO A
ELECTRONIC FAULT***

***WHEN TESTED THE UNIT
OPERATED INTERMITTENTLY
EXAMINATION OF THE CIRCUIT
BOARD REVEALED A DRY
JOINT AT THE “G” SWITCH***



***WITH CLOSER EXAMINATION OF
THE CIRCUIT BOARD IT WAS
FOUND THAT MODIFICATIONS
HAD BEEN DONE TO REPAIR A
DAMAGED SOLDER PAD***





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***THE HEAT TRANSFERRED WHEN
SOLDERING CAN EFFECT
OTHER PARTS OF THE CIRCUIT
EXTREME CARE MUST BE
TAKEN WITH REPAIRS***

***ADVANCES IN TECHNOLOGY
HAVE MADE IT POSSIBLE TO
ACHIEVE AMAZING FEATS IN
CONTROL OF FLIGHT
SURFACES, MONITORING AND
AIRCRAFT NAVIGATION***

***HOWEVER AGEING AIRCRAFT
AND NEW TECHNOLOGIES
PRESENT A CONSTANT PROBLEM
WITH THE AMOUNT OF TIME AND
MONEY REQUIRED TO KEEP
THESE SYSTEMS OPERATING AT
THEIR FULL POTENTIAL***

***SOME OF THE ELECTRONIC OR
ELECTRICAL FAULTS CAN BE
MAINTENANCE INDUCED:***

- ***REWIRING IN A NON-APPROVED MANNER***
 - ***UNPLUGGING AN ELECTRICAL OR
ELECTRONIC CONNECTOR***
-

***ELECTRICAL , ELECTRONIC
CONNECTORS AND PLUG
HOUSINGS CAN BECOME
DEGRADED BY TEMPERATURE
EXTREMES, VIBRATION OR
CHEMICALS***

***IT IS POSSIBLE TO PUSH A
VITAL CONNECTOR PIN
COMPLETELY OUT OF THE
CONNECTOR BY SIMPLY
RECONNECTING, RESULTING IN
INTERMITTENT, SHORT OR
OPEN CIRCUITS***

QUESTIONS ?

***THANKS FOR
YOUR TIME***
